AFGL-TR-77-0100



REPORT ON ATMOSPHERIC ENVIRONMENT INTERACTIONS WITH FREE AND TETHERED BALLOONS

ROBERT L. VESPRINI M. PATRICIA HAGAN

The Trustees of Emmanuel College — 400 The Fenway Boston, Massachusetts 02115



April 1977

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AIR FORCE GEOPHYSICS LABORATORY AIR FORCE SYSTEMS COMMAND UNITED STATES AIR FORCE HANSCOM AFB, MASSACHUSETTS 01731

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	AIR-LAUNCHED BALLOON SYSTEM		
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	THIS REPORT CONTAINS INFORMATION IN FOUR	R AREAS: (1)	
	AIR-LAUNCHED BALLOON SYSTEM (ALBS) DESCRIPTION, FLOWCHARTS, AND DOCUM		
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	IV BALLOON FLIGHT SIMULATIONS USING EGLIN AFB, HOLLOMAN, AND WHITE SANDS WIND DATA.
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INTRODUCTION

This report contains information in four areas:

(1)
Air-Launched Balloon System (ALBS): Programs, their description,
Flowcharts, and Documentations,

[2]
[4]
Balloon Tracking Using VOR: Description and Logic of Mathemas 'cal
Development, and a Program Employing These Techniques (Including
Flowchart and Documentation);

[5]
[6]
[7]
Tethered Balloon Experiment: Analysis and Recommendations;
[7]
[8]
Balloon Flight Simulations Using EUGIN AFB, HOLLOMAN, and
WHITE SANDS Wind Data.

I. ALBS Program Documentation

Introduction

This write-up describes programs pertaining to the Air-Launched Balloon System (ALBS), which were originally written for a desk-top calculator (Wang Model 452-1), by Andrew Carten (AFGL), ALBS development program manager. 1 These programs have now been put in Fortran so that they may be run on the the CDC 6600, and this section provides all information necessary to run these programs. For each program are provided a description, listing, flow chart, sample deck set-up, sample output, and definitions of all nomenclature, i.e., the variable names, used in the program. As much as possible, the original program names, subroutine names, and variable names have been retained. The underlying mathematics and logic, unless different, are not described in any depth For a more comprehensive treatment the reader should consult the Carten report.1

The first program, called CHUTE, treats the ALBS system as the drogue and main parachutes are being deployed. The second program, called P14B, treats the system as the balloon is being inflated.

Note that within the descriptions, flow charts, etc. variable names are often referred to without explanation.

The reader is referred to the appropriate section on 'Nomenclature'.

Program CHUTE

This program computes theoretical-performance data (velocity, drag, dynamic pressure, etc.) for the ALBS system starting from the point when either the drogue parachute or the main parachute is beginning to open until it is fully opened (0stsTF). The computations are based on an initially chosen opening time (TF=TFI). The volume of the filled parachute is then computed. If the calculated volume, using the initial opening time, is sufficiently close (less than 200 cu ft) to the theoretical volume, the program goes on. If not, a new opening time is chosen, and the first calculation is repeated.

When an acceptable opening time has been found, the behavior of the system at total computed. The calculations are continued until the velocity of the system is 'sufficiently close' to the equilibrium velocity (VEHI). The criterion for 'sufficiently close' is satisfied when the change in the dynamic pressure (Q) becomes 'very small'. The system data at this point become the starting data for another program not described here.

The program then reverts back to the time period during which the parachute was in the process of opening $(0 \le t \le TF)$, and it calculates deceleration and opening shock forces.

Data input to the program is by means of a READ statement. Each time the READ statement is executed, i.e., each line (or card) of input data is read a new set of calculations is done. Any number of sets of calculations may be done at one time. In this way, parametric studies of any of the parameters in the READ statement can be done easily. The data must be 'unformatted', i.e., the variables must be separated by commas.

Each time a new opening time (TF) is tried, subroutine
P4U is called with its first argument equal to 0, and one
page of output is written. When an appropriate opening
time is found, P4U is called again, but this time with its
first argument equal to 1, and calculations are made for
t>TF. This output is printed on the same page. Subroutine
P10 is then called to calculate the acceleration and opening
shock forces, and this output is written on a new page.

Tape 1 contains the output of main interest, while on tape 2 is wrinten supplementary information as well as results of calculations used for checking purposes.

(Note that in the program there are statements with 'CHECK' written in the first 5 columns. Because there is a 'C' in column 1, they are treated by the computer as comment cards.

In order to use these statements the 'CHECK' must be deleted.)

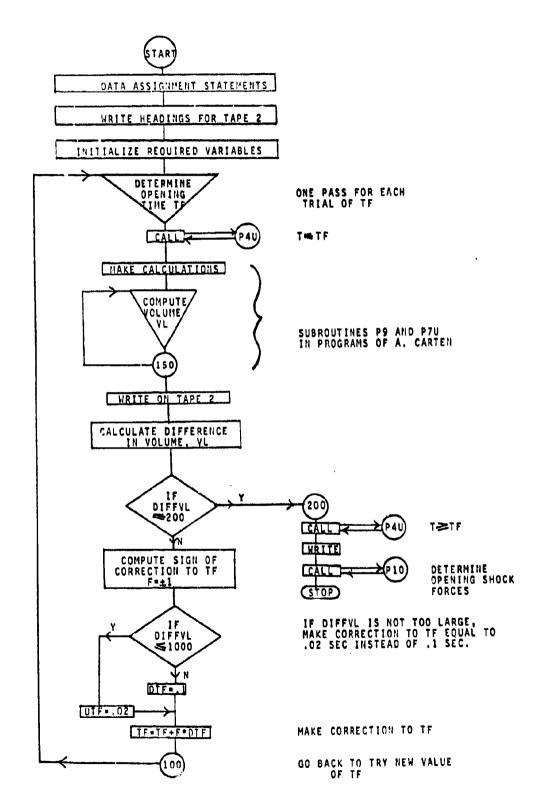
	PROGRAM CHUT	74/74	DPT=1	FTN 4.5.414	95/12/77	13.12.3
ı		PROGRAM CHUTE	(TAPE1, TAPE2, ENPUT, T	APE 5= INPUT) ONS E, 2/77 , DO, CP, 40	000100	
	CG	A	5-5-5-4-5-17 - 64-6-11 - 67-5	ANG	000110 000120	
	60 60	HATTICH FOR B I	CAPTEN. AFGI	042	003130	
5	00	SY ROBERT VESP	RINI. EMMANUEL COLLEG	E. 2/77	000140	
-	ČČ		4,		000150	
		COMMON VAELY A	(21), DVL(11)		000160	
		COMMON /CONST/	PI,G0,00,TF,G0,H		000170	
19		JATA P1,50/3+1	4199, 32.2/		000160	
4 "	c c	PEAN DATE			000200	
		READ (5.4) KM	AX. HO. W. TFI. d. CL	. Du. CP. 40	004213	
		IF (EOF(5) NE.	O) STOP		007550	
_			FOR TAPEZ UATE(TODAY) W,D0,CD,CP UIRED JARIABLES = TFI J/L+PI++2) +D0++3		000520	
13	¢c	HRITE HEADINGS	FOR TAPEZ		000540	
		MRITE (2,20)	DATE(TODAY)		04 2 00 0 C3 2 0 0 0	
		-2116 (5+26)	M, DU, CD, CP		000540	
		M471F (5)541			000260	
20	CG	INITIALIZE REO	UIRED JARIABLES		000890	
		F + 1. \$ TF	= TFI		000300	
		37F * +1			000310	
		50 m (P[+D0++8	1/4.		000350	
25		ACHAX . (5')(2	***************************************		000340 332343	
6 7	26	START (AL CHILAT	TONS, ONE VALUE OF TE	FOR EACH KOUNT	000350	
		SO 100 KOUNT		TON BROW NOOT!	900360	
		CALL OLD IN.HO	M. VII. VF 1 . H.)		000370	
		VIHI = SORT(W/	(.001189+SIGM4 (HO)+C	o+ 50))	000300	
3)	4.0				000390	
	CC	AT . 0.	AND PPU IN A. GARIE	N'S PROGRAMS	000400 000410	
		31 * D0** 2*TF/	91		10:420	
		00 191 J = 1		·	000430	
35		T = FLOAT (J-1)	/10.		000+40	
				.) - 2.*GP=T+(L,-T))	0 0 0 4 50	
		VL = VL + .1*0) AF (1)		69 4 20 6	
	141	CONTINUE	•		003470	•
4)		WALLE (5150) A	Wi		000490	
•			•••		300563	
	SC	IS TE ACCEPTAB	LET IF IT IS, SO TO	200.	000510	
		DIFFYL . ABS (V	ILMAX - VL)	•	000520	
4. •		IF (DIFFVL.LE.	200.) GO TO 200		003 > 30	
4 3		F # (VLMAX = V	/L)/JIFPYL /MINT.TR.V:MAY.U1.318	ru:	0005 ±0	
		watte (2.18) w	COUNT. TF. VI MAY. JA . DIF	FJI.	000540	
		IF (DIFFYL.LE.	1000.) DTF = . 02	• • •	003 9 70	
		TF # TF + F*D1	F		000583	
çı	101	SOUL THRE		200. FYL FYL	000>90	
	201	SONT INDE			000600 000510	
	20	MAKE CALCULATI	TONS FOR TIME.GT.TE		000620	
	,,	CALL PLU (1.H.	HF. VF1. VF. d)		001633	
55					000640	
			the address actions an		40000	
		MAILE (5,10)	KOUNT, TF, VLMAX, VL,JIF	FVL	000660	
	ac	HAVE CALC OF	ACC AND BORCES. HUT) &	CHUTE IS SEING DEPLOYED.	003679	
6.0	40	CALL PIO (HO)	ACC AND FURUES! HOLLE	CHOIC IS GEING DEPECTED!	000430	
•••					000700	
	10	FORMAT (1HD, I		,	000710	
	11			= *,F8.2,8X,*VEHI = *,F7.2	000720	
	**			LC VOLUME = 5,F10.2)	000730	(
65	\$ 0 \$ 2			INFORMATION*,10x,410; OT LISTED ELSEWHERE*,//	000740 0007 9 0	•
	£ E			IAN OF CHUTE . * FB.3,//	000760	-
			RAG - *, F6. 3,13K, C		000770	
<u> </u>	24	FORMAT (1HO, 1	OX. FOTHER DALGULATED		000780	
70	26	FORMAT (1HO,1	1F1 0.23		. 000790	
	•	STOP			00000	
		END			000813	

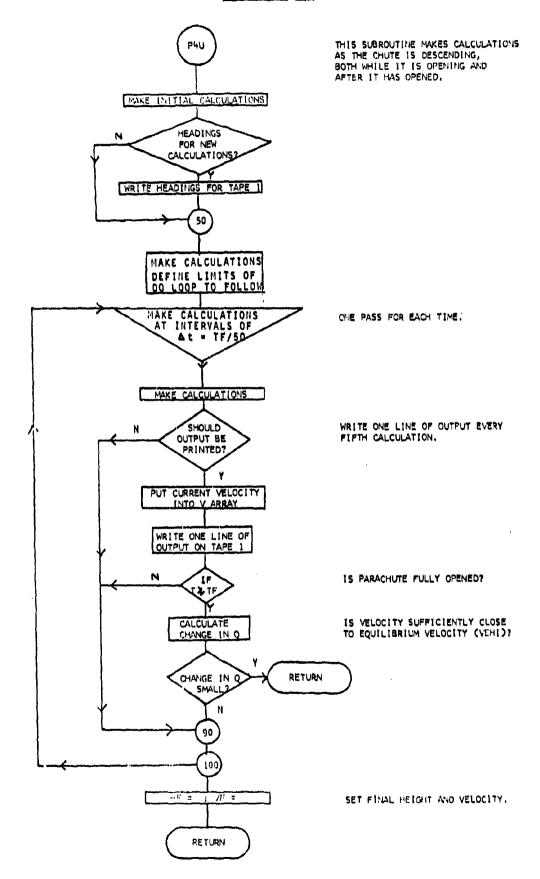
SUBRO	UTINE P10	74/74			FT4 4.5+414		12.12.38
1		SUBROUTINE PERSONNELS COMMON /VEL/	10 (H) V(21), DVL(1 [/ PI,GO,OO,T	1) F,CD, H BLES +G0+U0++3) VALS OF TF/10.		0	
5	cc	INITIALIZE RE	:QUIRED VARIA	BLES		001560 001570 001560	
10		A = (W*10000) B = 120. +CDS: C = A+50).)/(SIGMA(H) DM/D0++3	#G0+U0**3}		001590 001500 001510	
	CC	HRITE HEADING HRITE (1,1)	SE ON TAPEL DATE(T COAY)			001620 001330 001643 001683	
15	GG	START CALCULA DO 100 I = T = FLOAT(I=	TIONS, INTER 1,11 1)/10.	VALS OF TF/10.	1446 4 95 567	0016 50 0016 70 0016 50	
5 0				TF+T+V(I)++2 + C+TF			
25	100 30	ARITE (1,10)	T, DVL(I), D	VOT(I), F, F0, G		001/90	
••	CC	HRITE ACOITI HRITE (1,20)	OHAL INFORMAT A, B, C, CDS	I CN		001770 001780 001790	
30	1	FORMAT (1H1, 1,19X,**********************************	10x,*PROGRAM DVOL (CUFT ,*F0/H (G)*,/	P9*,13X,*PKOGRAM P:) DV/OT (FP\$/5)*,1	DX, *F (LB) * , 9X ,	// 001800 001810 001820 001830	
35	\$0	FORMAT (1HO, 1,8X,+CDSO # RETURN	10x, 4A = 4, F 4, F12, 2)) -2;F14:3) 12:3;5X;PB	,5x,+C = *	,F12.3,//001840 001840 001860	•
		END				001870	

	FUNCTION SIGN	14 74/74	OPT=1	FTN	4. 4 + 4 1 4	05/12/77	13,12,36
1		FUNCTION SIGN	(H)			001880	
		DIMENSION SIG	(25)			001690	
	\$	THIS FUNCTION	INTERPOLATES BETHEEN	VALUES OF THE D	ENSITY RATIO,	001900	
	C	SIVEN HY SIG	IN INTERVALS OF 1360	FT.		601913	
-		DATA SIG/.971	07942781513688	11 88 170 83390		001929	
			9 76206, . 73 859, . 7186		•	001930	
			661 42 1, . \$8 946 , . 0702			0019-0	
		3 51534 4979	8 48108 - 48462 448	9/		CO 1 9 85	
		J = H/1000.	• • • • • • • • • • • • • • • • • • • •			. 001960	
10		XJJ = H = 100	D. * FLUAT (J)			001970	
		SIGHA * SIGLJ) + (SIG(J+1) - SIG(J	.0001\LLX# ((001980	
		RETURN				001993	
		ENO				002001	

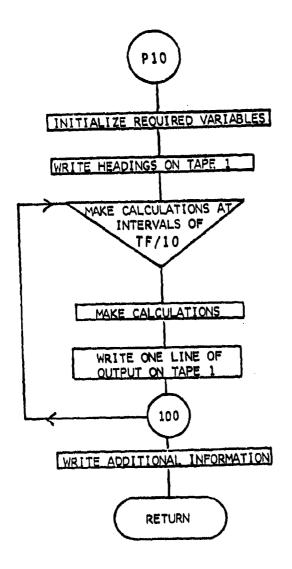
SUAROU T I		74/74	0PT=1		TN 4.5+414		13.12.35
1 5	00	SUBROUTINE PA IFLAG = 0 IFLAG = 1 COMMON /CONST COMMON /VEL/	U (IFLAG, HO, HF, CHUTE IS BEING CHUTE HAS BEEN T/ PI,GO,DO,TF,C V(21), CVL(11)	VO, VF, B) OEPLOYED DEPLOYED O, M S NECESSARY. CO+SO+FLOAT (IFLAG)/ IBB + 49 S OF OT = TF/50.		003 8 20 3 00 8 30 0 0 0 8 40 0 0 0 8 50 0 0 0 8 60	
	cc	INITIALIZE RE 50 = (FI+D0+4 2 = .001189+	EQUIRED VARIABLE P2)/4. Sigma(H0)*V0**2	s		000 6 70 000 6 80 000 6 90	
10	С	WRITE HEADING IF (IFLAG. EQ. WRITF (1, 10)	GS FOR TAPEL, IF .1) GO TO BU DATE(TODAY)	NECESSARY.		000910 000920 000930 000940	
15	50	WRITE (1,2) CONTINUE V(10+IFLAG+1)	9, Q, VO, HO) = VI = V1 = V6	ı		000 9 53 000 9 60 000 9 70 000 9 80	
2 0		H = M0 3I = 0. DT = TF/50. DV1 = DV0 = 188 = 50*IFL	-G0*DT*{1 Q*[AG + 1	33*S0*FLOAT(IFLAG)/	w)	300 993 00 10 30 00 10 10 00 10 20 00 10 33	
25	ÇC	START CALCUL DO 100 I = I = FLOAT(I) TIME = TTTF	ATIONS, INTERVAL 188, IFF /50.	.S OF DT = TF/50.		001040 001050 001060 001373 001080	
3 0		DYPRE = 2.40 YPRE = V1 + DHPRE = VEAV DHPRE = JEAV	V1 - DV0 DVPRE VPRE)/2. +DT STGMA(H+ DMP2F)+	, ,pp:0+2		001100 001113 001120 001130	
35		5 = Q*CO*A SVI = -GQ*OT VI = V1 + DV SMI = -F*{VI H = H + DHI	* (1 D/H)			001180 001160 001170 001140 001140	
4 9		V1 = VI DV0 = CV1 DV1 = CVI				001200 001210 001220 001233	
L 5	70	#RITE ONE LI II = MCO(1,5 IF (II.NE.0) IJ = I/5 + 1 V(IJ) = VI #RITE (1,1)	NE OF OUTPUT, E , GO TO 90 ,T,TIME,Q,OVI,V	I 188 + 49 S OF DT = TF/50. VPRE••2 VERY FIFTH CALCULAT I,U,H HEN DIFF IN Q IS SH	TON	001240 001250 001260 001270 001270	
50	CC	FOR IFLAG = IF (IFLAG.EG) DIFFO = ABS(IF (DIFFO.LE)	1, CALC STOPS M 1.0) GD TO 90 Q - QI) 001) RETURN	MEN DIFF IN Q IS SM	IALL	001310 001310 001320 001330 001340	
5.5	90 JHEDK	CONTINUE IF (IFLAG.EG	1.1) HHIFE (2,1) A,T,TIME,Q,OVI,		301360 001370 001380	
€ 0	0HE0K 190 30	CONTINUE		OVI, VI, D, H . SENT BACK TO ALBS		001330 001400 001410 001420 001430	
65	1 2 10	HRITE (2,2) FORMAT (1HO FORMAT (1H	8,Q,VF,4F ,10X,F11.2,F8.1, ,10X,F11.2,17X,F ,10X,*PROGRAY P4	T(SEC) Q(PSF)		001440 001440 001440 001470 001470 V(F>5)001440	
70	•	FETURN END				001900 001510	

PROGRAM CHUTE





SUBROUTINE P10



THIS SUBROUTINE COMPUTES OPENING SHOCK FORCES.

ONE PASS FOR EACH TIME INTERVAL.

```
end of file
10.23771.1430,3.5,962.11,.755,72.945,.048,-47.62
    program '
 PROSPAN CHUTE (TAPEL)TAPES) INPUT)TAPES=INPUT)
      end of record
COPY, TAPES, OUTPUT.
COPY, TAPE1, DUTPUT.
REUIND. TAPEL, TAPEZ.
LIGO.
LDOET, PRECET=ZERO.
ATN.
VESPA.
             2037
               MAHIFIELD
 -11
               1 111 1
```

PROGRAM PLU 35/12/77

		•		233.64	26451.65		3.50 26217.80
23628.28	1512.35	-30.22	4282 *	.5110	3.50	1.0	4179.08
23639.04	1641.74	-31.74	.3337	. 5637	3.15	6	3857.38
23650.+5	1677.35	-33.52	.3899	.6284	2.80	π Ω •	3535.69
23662.54	1718.79	-35.60	.4552	.7883	2.45	.7	3213.99
23675.41	1762.94	-38.01	.5248	. 8673	2.10	•	2892.29
23689.19	1880.58	-40.75	.584€	.9280	1.75	r.	2570.59
23703.97	1814.04	-43.75	.5053	1.0634	1.40	•	2243.90
23719.81	1768.89	-46.53	.5342	1.2157	1.05	M)	1927.20
23735.58	1620.04	-4·5.97	•2996	1.3365	.70	• 5	1605.50
23753.90	1334.27	-49.65	1589	1.3765	.35		1283.81
23771.00		-47.62		1.2623			962.11
H(F1)	U(LB)	V(FPS)	BY(FPS)	QUESTI	r (SEC)	1/TF	area (sqft)

2	Đ0	00	7.8	18	35		16	14	11	92	98	9	53	M	59	55	9.
H(FT)	23771.00	23754.00	23135.78	23720.18	23784.35	23689.65	23675.94	23663.14	23651.11	23639.76	23628.98	23618.64	23688.53	23598.53	23588.59	23578.57	23568.76
0(18)		1333.73	1519.89	1769.51	1315.40	1802.79	1764.90	1720.70	1579.09	1643.27	1613.69	1516.01	1471.37	1450.61	1439.90	1435.07	1432.76
V (FPS)	-47.62	-49.54	-46.97	-46.70	-43.77	-48.78	-36.83	-35.62	-33.54	-31.76	-30.23	-29.38	-28.85	-28.64	-26.53	-25.48	-28.45
DW(FPS)		1509	.2976	.5 321	.6848	.5842	•5249	•4556	9862*	.3342	.2879	.1348	. 1648	.0314	.8155	6400.	.0043
9(15)	1.2623	1.3760	1.3364	1.2161	1.0692	. 3283	.8882	.7891	.5290	.5642	.5114	. 48 85	.4663	.4595	+656+	. 4543	.4541
T (SEC)		.35	.70	1.04	1.39	1.74	2.19	2.44	2.78	3.13	3.48	3.83	4.16	4.52	4.87	5.22	2.57
1/15		•		M.	•	ın	•		€,	6.	1.0	1.1	1.2	£ • 3	1.4	1.5	1.0
AREA (SOFT)	952.11	1283.61	1685.50	1927.20	2248.98	2578.39	2892.29	3213.99	3535.69	3857.38	4179.08	4179.08	4179.08	4173.88	\$173.U8	£179.05	4179.38

26310.60

CALC VOLUME =

26217.88

VOLMAX =

OPENING TIME =

26.53

VEHI =

17

ã.	PROGRAM P9	&	-	PROGRAY P10	910		95/12/77	
	T/TF	<u>.</u>	FOL (CUFT) 04/6T	19//0	(FPS/S)	F(LB)	FB(L3)	F0/H (6)
	υ· Θ		-0 - 66		68.19	-870.26	55 3. 74	3.94
	•1		9694.18		29,39	-375,05	1854.95	7.38
	•2		22572.17		•26	-3,38	1+26+62	0 0
	M)		33143.38		-17.13	218.55	1640.55	1,153
	.		39675.81		-25.39	324.85	1754.85	1.227
	ň	-	41929.29		-28.13	358.96	1788.96	1,251
	, P	~ ~	48 212 . 89		-28.87	358.28	1788.20	1.250
		1	34916.18		-26.84	342.49	1772.49	1,240
	•	••	26323.58		-25.26	322.33	1732.33	1.225
	6	~	14647.51		-23.69	302, 31	1732,31	1.211
	1.0		-0.00		-22.26	264.11	1714.11	1.193
≪	H	24.458	6 6 2	11	.97549	# cs	756.573	
COSO	**	3155.28	5		,			

PROGRAM P9

AL3S -- SUPPLEMENTARY INFURNATION 05/12/77

INPUT VARIABLES NOT LISTED ELSEWHERE

72.95	
11	n
DIAM OF SHUES	C OF POROSITY
1431.0	.755
H	11
SYSTEM WEIGHT	COEFF OF DRAG

	-38.22		-0.9		-38.23			
	-31.7.	0.00	14724.62	•	-31.76	3.6	1-647-51	
23628.20	-33.52	8	26458.84	23624.38	-33.54		26.323.58	
	-35.60	:	35851.80		-35.62	. 6	34-318-15	
-30.22	-38.01	:	58-12-02	-38, 23	-36.63	:	48212.89	
	-48.76	•	22782.97 33327.76 39888.51 42148.76 40428.35 35891.88 26458.84 14724.62		8.76		22572.17 33143.38 39675.31 41923.29 48212.89 34318.13 26323.50 1.647.51	
.5118	-43.75		19888.51	233.84	-13.77	:	39675.31	93.11
ABLES	-56.69		33327.78		-46.78	:	33143.34	
ATED VARI	-44.97		22782.97	7.88 2645	-48.97	•	11-21522	7.88 2631
OTHER CALCULATED WARIABLES 962.11	-49.65		9751.55	3.58 26217.88 26451.65 962.11	-49.64		9694.18	3.48 26217.88 26318.88
6	-47.62	0 - 0	2.		-47.62	6 . B	7	~

Nomenclature

Program CHUTE	
В	Cross-sectional area (b) of parachute at t=0
CD	Coefficient of drag, CD
CP	Coefficient of porosity, Cp
C1	$(C1) = d_o^2 t_f / \pi$
DIFFVL	Absolute difference between the calculated and theoretical parachute volume
DTF	Correction to tf to get DIFFVL below an arbitrary limit
DVL	Rate of change (dV/dT) of volume with $T = t/t_{f}$
DO	Parachute diameter, do
F	Number used to decide whether $\mathbf{t_f}$ should be increased or decreased to produce an acceptable parachute volume.
GO	Acceleration of gravity $(g_0 = 32.2 ft/sec^2)$
Н	Final height returned from subroutine P4U
HF	Final height after chute has achieved equilibrium velocity
НО	Initial height, H ₁
J	Index used to increment successive iteration
KMAX	Maximum number of different opening times (t_f) that will be tried for a given run.

This is an additional safeguard against the program going into infinite loop.

KØUNT	Index used to count the number of different opening times (t_f) that have been tried.
PI	π
SIG	Array defining σ at 1000-ft intervals, where SIG(I) = σ (h=1000*I), up to 25000 ft.
SO	Cross-sectional area (S_0) of parachute when fully opened $(t=t_f)$
T	Fraction of opening time, T = t/tf
TF	Opening time, tf
TFI	Initial value of opening time, read into the program
TØDAY	Dummy variable used to print date
v	Array containing values of velocity (v), where $V(1) = V(T=.1), V(2)=V(T=.2)$, etc.
VEHI	Equilibrium velocity $(V_{e(Hi)})$ of parachute system
VF	Final velocity, VF=VBHI
VFI	Velocity after parachute is fully opened $(t=t_f)$
VL	Parachute volume calculated using opening time t=tf
VLMAX	Theoretical parachute volume
vo	Initial velocity of parachute system (at t=0)
W	Weight of parachute system, W_{S}

Nomenclature

Subroutine P4U

A Cross-sectional area of parachute at a

given time t

B Same as in CHUTE

CD Same as in CHUTE

D Drag of parachute, D

DHI Change in height in time DT=tf/50

DHPRE Initial calculation of change in height

DIFFQ Change in q in time t=.1tf. This is used

only after the chute is fully opened
(IFLAG = 1) to determine if the velocity
(V) is close enough to the equilibrium

velocity (VEHI)

DT Time increment/iteration; DT = $t_f/50$

DVI Change in velocity (V) in time DT

DVL Same as in CHUTE

DVPRE Initial calculation of velocity change

in time DT

DVO, DVI Used in determining DVPRE. They are the

two previous DVI's

DO, GO Same as in CHUTE

H Height

HF Final height when P4U returns control to

CHUTE

HO Initial height when P4U starts

I Index used to increment successive iterations

IBB, IFF	Beginning and final values of index I
IFLAC	Index used to tell P4U whether the parachute
	is in the process of opening (IFLAG = 0 ,
	$0 \le t \le t_f$) or is fully opened (IFLAG = 1, $t \ge t_f$)
II	Index to determine whether a particular
	iteration should be printed. (Only 1 out
	of 5 is printed)
IJ	Index used to place a velocity value (VI)
	into the array V; this index corresponds
	to a time such that $V(IJ)=V(T)$, $IJ=10*T+1$
PI	Same as in CHUTE
Q	Dynamic pressure, q = 4ev2
QI	Previous value of Q; used to calculate DIFFQ
SO, T, TF	Same as in CHUTE
TIME	Real time t(sec)
TØDAY, V	Same as in CHUTE
VEAV	Average velocity over the interval DT
VF	Final velocity when P4U returns control
	to CHUTE
VI	New velocity at end of time increment DT
VPRE	Preliminary value of now velocity at end
	of time increment DT
vo	Initial velocity when P4U starts
VI	Used to store the previous velocity (VI)

Nomenclature

Subroutine P10

A,B,C Coefficients used to calculate DVDT

CD Same as in CHUTE

 $(C_{p}S_{o})_{m} = C_{p}S_{o}$

DVDT Acceleration of parachute system (ft/sec²)

DVL, DO Same as in CHUTE

Force produced by the drag of the

parachute (1bs)

FO Sum of the drag of the parachute and the

force of gravity

G Acceleration in g's corresponding to FO

Height at which o is to be calculated

This value $\sigma(H)$ is held constant

throughout the subroutine

I Index used to specify the time (Tw.1*I)

PI, T, TF, TØDAY, V, W Same as in CHUTE

Function SIGMA

SIG

Н Height at which SIGMA is to be calculated Index used to determine 1000-ft

interval

Same as in CHUTE SIGHA Value of $\sigma(H)$ calculated and returned

to calling program or subroutine

Variable which is used to interpolate XJJ

 σ between h=1000J and h=1000(J+1)

Program P14B

This program treats the ALBS system starting from the point when the balloon has been completely extracted, and the helium has begun to enter the balloon. Calculations are done every 200 ft, as the system descends through the atmosphere, and one line of output is printed for each height. The effect of the system weight, the drags of the drogue and main parachutes, the lift produced by the increasing volume of helium in the balloon, and the change in temperature and pressure of the atmosphere are all taken into account by the program.

Data is introduced to the program thru a READ statement. Each time a card of data is read, another set of calculations is done. Any number of sets of calculations may be done during the execution of the program. In this way, parametric studies of any of the parameters in the READ statement can be done easily. The data must be 'unformatted', i. e., the variables must be separated by commas.

The output of the program is written on tapes 1 and 2

(See Figures 1-4). After the headings are written, one
line of output is written for each height (200-ft intervals).

Since the values of so many variables are written, the

width of one page of computer paper is not enough to write

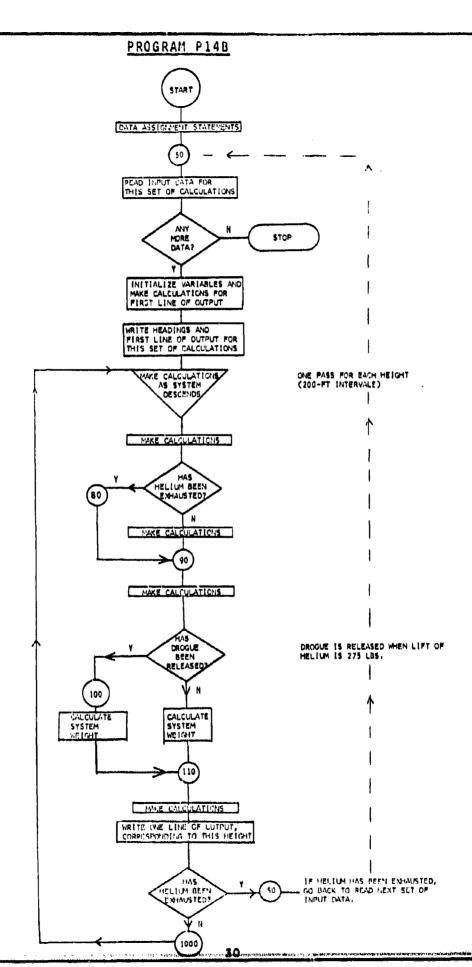
them all across the page in one line. Hence, for each height, some of the variables are written on tape 1 (Figures 1 and 3) and the rest on tape 2 (Figures 2 and 4). After completion of the job, the pages are separated and each set of calculations is taped together so that the data for a given height can be read across the page in one line.

Each time another READ statement is executed, another set of calculations is done. For each new set of calculations, the headings are reprinted at the tops of new pages on tapes 1 and 2.

The Fortran version, explained here, also contains two new features not found in the original version written for the desk-top calculator. The first takes into account the adiabatic heating of the gas in the balloon as the system descends into a denser and denser atmosphere. The second provides an interpolation to determine the exact height at which the helium is exhausted (instead of the height of the nearest 200-ft interval) and the values of the other parameters at that height. These features are explained in detail below.

	PROGRAM P143				05/12/77	13.12.45
1		PROGRAM P148	(TAPE1, TAPE2, IN	PUT, TAPES= INPUT)	000100	
	CC				006110	
	3 C 3G		INFLATION SALCU 1. CARTEN, AFGL	CALIONS	000120 000130	
5	33	TY PORFET VES	RPRINI. EMMANUFL	COLLEGE, 2/77	000140	
•	čč	21 100011 000	, Kanaj aminanoaa	00011011 1777	J051 9 3	•
		CIMENSION PRE	E88(29), SI3(25)		000160	
		DATA PRESS/.9	9643,.9298,.8962	,.8636,.8320,.5014,.7717,.7428,	000170	
		1 .7149,.6878,	6616, . 6361, . 61	,.8636,.8320,.8014,.7717,.7428, 18,.8876,.5646,.5422,.5206, 27,.4081,.3880,.3716/ 3,.88811,.86170,.83890 ,.71568,.69333,.67151 ,.87021,.88144,.83318	000180	
10	ì	2 .4997,.4799,	,,4999,,4410,,41	27,.4051,.3850,.3716/	000193	
			1079:74670;:7121 No7698£. 71486	3,100511,1001/U1153PYU	000200 000210	
		266022629	46 f09215844f	,, 471900, 40 93 33, 4 OF 151	000210	
	·	3 51 534 497	98 481 0 8 464 62	,.446 59/ 3,-200./	003230	
17		DATA PI.CI.DI	H /3.14189.3.841	3,-200./	000240	
	2033	WEAD DATA			000280	
	3.0	READ (S.4) H	I, WSREF, THE, 1	R, TREF, VO, DT, TLREL, HO, DAD,	DAM 000260	
		IF (EDF(5).NI	E.O) STOP		00:5 10	
					000890	
S 0	0000	THUE # 102.4	EQUIRED VARIABLI	.5	000290	
			YEMP = THE		600310	
		IFLAS = D S		•	000350	
		ITOP = IFIXE	•	-	000330	
25			124) ITOP . 124		006340	
		I = ITOP + 1			000350	
		J • I/#			000360	
		77 * 1 - 247			000370	
		SIGMA = SIG (J) + (SIG(J+1)-	SIG(J))+FLOAT(JJ)/B. PRESS(J))+FLOAT(JJ)/B.	003360	
30		P = PRE53(J)	+ (PRE35(J+1)=	RESS(J))*FLOAT(JJ)/B.	000390	
		JOL = TM = T	M U.		988488 98848	
		JA3 - UNU T	VAT		003423	
	3688	WRITE HEADING	GS AND DATA AT 1	NITIAL HEIGHT	000430	
34		WRITE (1.10)	DATE (TODAY)		000440	
		HRITE (1, 11)	THE TREF TROOT		000450	
		HRITE (2,12)	THE.TREF.TR.DT	LREL	000460	
				•	000470	
40		WRITE (2,2)	MT. STOMA. D. I	. .	000480 000490	
- 0			HI, SIGMA, P. HI. WSREF. DAS		000530	
			-,	•	000510	
	0000	START CALCUL	ATIONS, 200-FT	INTERVALS	000520	
		DO 1000 II	# 1,ITOP		000530	
45		I = ITOP - I	I + i		000340	
		J * 1/5		. , \	00122	
		DY . DTTR		•	000560	
		THM = TM + D	м		000570	
e n	0000		CEN EVMANCTERS	TF NDT. GO TD 60.	000560	
			MHE) SO TO SO	Er 1901 9 90, 10 004	000570	
		IFLAG = 1	90 10 90		000610	
		F = (THHE-TH) / DM		023953	
		TH = THHE	•	•	000630	
55		30 to 90			000840	
	9.0	TM = TMM			000550	
	30	CONT INUE			00:061	

	PROGRAM PINA	74/74	0PT=1		FTN 4.5+414	05/12/77
		H = -DH+(FLO	AT(1+1) - F)			300 5 73 000 6 80
F 0		XJJ = -DH+(F				000690
		TAIR . TREF	- 2.* (H-HI)/	1000.	440.10	000700 004 7 10
		SIGMA * SIG(J) + (5[G(J(1000. 1) - SIG(J))*XJJ 1) - PRESS(J))*X TEMPL + OM*THE)/T	/10U0 •	000/20
		TEMP # ((TH	DM) * (TEMP+ D)	ENPL + DHOTHE) /T	M	000730
€ 5				EMP))/(CL/TEMP)		000740
		TL = THPRLM				060783
					NOT, 30 TO 100.	000 76 0 000 770
	cccc	IF (TL.LT.TL			NOI, 30 13 1881	000780
70		HS . HSREF -		***		606740
		DAD = 0.	_			000800
		30 TO 110	•.			000810
	100	MS & MSREF -	· TL			030520 001 8 30
7 5	110	CONTINUE				000840
• •		VOL = (TH+TE	(MP) / (C14P)			000680
		DIAB = ((6./				000660
		DAB + (PI/4.		. 5		006876
		DAS = JAB +				000880 000890
80		VEAV = (VE		189*SIGHA*DAS)))		000010
		31 . F+3H/VI				000910
		T = T + DT				000920
		VO = VE				000930
4.5		0 = .001189		SHAD = NO 4		0009 4 0
		DS = D8 + D		P 04 - 04444		000960
		HH # HS - D!				000973
						000760
90	0033	WRITE ONE L	THE OF DATA			000990 001000
		WRITE (1,3)	Mg Sig Mag Pg T A - W. Ti. Ms. ('Ai	,DIAB DAS, UAB, DB	E. Q. TH. TEMP	001010
						001020
	6683	IF CALCULAT	IONS ARE DON	E, GO BACK TO RES	AD MORE DATA.	001030
45		IF (IFLAG.E	0.11 GU TO	50		001040
	4020	CONT THUS				0010 5 0
	1000	CONTINUE				001078
	1	FORMAT (1HO	,* HEIGHT	SIGMA(AIR) P	RESS(ATH) T(AIR) *	, 001080
100		LTX, +L/H+BX+	D(TIHE)	TIME VEL (EQ	uil) *7x,1Hi,7x,4 mass (he)*, 001090
	_	S F TEMP (H			METANON MOLENALISE	J 011 J3 0 01110
	2		,* HEIGHT	TOTAL LIFT SYS (CDS 0) BALL	WT(3Y3) VOL(BALL)*	001120
		2 PURAG(DROG)				001130
105	3	FORMAT (1HO	, F7.0, F14.5;	F12.4, F12.1,3F12	. 3,F12,2,F12.4,F12.3,F1	2.11001140
	4	FURHAT (1H8	, F7 . 0 , 4 x , 2 F 1	. 2. 2, F 12. 3, 6 F L 2 . 2	,F12.3)	001150
	10			ON INFLATION CAL	CULATIONS+,/,	0011 60 0011 70
	11	1 FOPROGRAM	- 1 4 7 * 6 LE MB ()	II) FROM DEWER	3H = .F7.1.24X.	001180
110	••	1 TEMP (AIR)		* 1F7 - 11 / 1	3H = ,F7.1,24X,	001190
				[*6x,2Hm ,F11.4,2	QX,	001500
		3 * INITIAL D	(TIME)	· * , F9 . Z , / / / }	8.UITE - B PS 3	001210
	1 2	FORMAT (1H1		+DRAG AREA HAIN = +,F9.2,/,	GIRCLE TAPPACA	001530 001530
115		•			L	001240
4 4 7						0075 20
	11	FORMAT (1HO	.F7.0. EX.F1	2.5,F1 2.4,46x,F12	. 21	. 001560
	14	FORMAT (1HO		12. 2,24x, f 12. 2,24		001270
120		51 OP			·	001280 001290
753		EN O				2016 30



BEST AVAILABLE COPY

00179 P 0 5 101

```
end of file
22-00-1367,260,.34-1,260.2,-39.34,5.1,275,66,442.34,1080.65
22400,1267,255,.2441,260.2,-39.34,5.1,275,66,442.34,1080.65
22400,1247,250,.3441,248.2,439.34,5.1,275,66,442.34,1880.65
          program
   PROGRAM PIAB (TAPE1, TAPE2, INPUT, TAPE5=INPUT)
             end of record
COPY, TAPES, OUTPUT.
COPY, TAPE1, OUTPUT.
RELLINIO, TAPE1, TAPES.
. الآقال
LDSET, PRESET=ZERO.
CTH.
VECPA.
                                2037
                                     MANSFIELD
                                      1 111 1
ីវីវៈនិនិនិនិនិស្សាល់ នៅក្នុងប្រជាពីក្រុងប្រជាពីក្រុងប្រជាពីក្រុងប្រជាពីក្រុងប្រជាពីក្រុងប្រជាពិក្រុងប្រជាពិក្រុ
```

ALBS BALLOON INFLATION CALCULATIONS
PROCRAM PILE B5712777

. III	TEMP(HE) FROM DEWER = 258.8	.3441		TEMP(AIR)	TENP(AIR) INITIAL OSTINED	2.88.2			
ı		•					•		
PRESS(ATH) T(AIR)	8		5	D (TINE)	TIME	4EL (EQUIL)	o	MASSCHE	TEMP (ME)
.4157						-39.3%			
.4192 288.6	٠.	16	5.951	5.130	5.136	-38.54	. £737	1.755	8.852
.4227 266.0	_	~	5.951	2225	11.356	-38.85	. 8573	3.522	256.4
.4264 - 261.4	_	w	5.951	5.233	15.6+6	-37.68	.8427	5.319	9-8-52
.4368 261.5		IŃ.	5.952	5.351	21.996	-37.16	.6291	7.139	251.2
.4337 262.2	•.	W	5.952	5.412	56.489	-36.74	. 8161	d. 90B	251.8
.4373 262.6		in	5.952	5.473	31.642	-36-34	. 66 37	10.642	252.6
.4418 255.8		w	256.5	5.534	37.416	-35.94	. 7916	12.726	252.4
. 6448 263.4		50	5.952	5.595	43.811	-35.55	.7799	14.638	252.1
.4486 253.8	_	N.	5.952	5.656	46.566	-35.17	.768	16,555	253.1
2.4523 256.2		ıń	5.952	5.717	54.384	-34.79	.1572	18.581	253.5
.4561 261.6		5	2.952	5.779	64.163	-34.42	1942.	. 28,468	253.9
. 1539 265.E		10	5.952	5.341	66.114	-34.05	. 7353	22.457	754.3
**638 265.*		5	256*5	5.984	71.908	-33.69	.7245	24.467	1.452
.4677 26.8		5	5.952	5.968	77.875	-33,33	.7139	26.438	255.1
.4717 266.2		W	5.952	6.833	63.919	-32.37	,7834	28.552	4*552
.4755 266.5		IA.	2.952	6. 198	90.017	-32,62	• 6930	30.628	255.8
.4795 267.8			5.952	491.9	94-171	-32.28	.6827	32.726	256.2
*************		IÁ	5.952	6.232	182-483	-31.32	•6725	34.6-7	9-952
.4876 261.3			j. 951	6.311	166.764	-31.57	.6623	36.992	6*952
2·102 5164.		ľ	5. 951	6.37t	115.074	-31.22	5539°	39.160	257.3
. 4557 266.6		Ŋ	5.951	144.9	121-515	-31.88	. t421	+1.352	257.7

16886.	.57871	1669.	2 69.0	5.958	6.513	120.129	-36.53	.6321	4.3.568	258-8
17886.	.57436	er 36 .	1.652	5.958	6.567	134.616	-36.13	•6221	45.618	7.852
17681.	.57791	.5861	2 (3. 8	5.949	6.239	141.985	-33.41	.7669	43.076	258.6
17486.	921F5*	.5122	278.2	5.9.3	6.823	146.928	-33.61	.7537	54.246	2.632
17288.	.50501	.5164	278.6	5. 23	969.9	153.822	-32.53	.7412	52.313	5-652
17666.	.58346	.5216	271.0	156.5	6.166	153.196	-32.25	.7266	54.418	268.0
16888.	.59341	. 5249	1.12	5.951	6.239	165.427	-31.86	. 216	55.532	266.4
16688.	.59736	2625.	271.8	5.951	6.314	171,741	-31.46	.7848	56.678	260.8
16488.	.68131	. 5336	272.2	5.951	6. 391	170.122	-31.10	.6917	64,851	261.1
16256.	.66526	5373	272.8	5.951	694.9	164.682	-34.72	.6794	63.858	261.5
16888.	.68321	2245	273.0	5.951	6.550	191-152	-34.35	. 6676	65.276	261.9
15686.	.61326	Z.	273.4	5.951	6.632	197.784	16.62-	. 6547	67.538	262.3
15636.	.61731	.5512	273.8	5.951	6.717	284.581	-29.56	.6424	69.812	7.292
15-88.	.62,36	. 5556	2.4.2	5.958	6.8%	211.305	-29.28	. 6381	72.124	263.8
15211.	19529*	.5681	27.6	5.949	6.893	218-199	-29.82	.6177	74.465	263.4
15886.	98629*	. 9846	2 15.0	5.949	586.4	225.164	-28.44	. E854	76.637	263.8
14881.	.63361	2695*	275.4	5.94	7.886	232.264	-28.85	. 5938	79.241	264.1
14686.	,63776	.5736	275.8	5.948	7.178	239.4K2	-27.57	.5636	61.677	264.5
14488.	26149*	¥976.	276.2	5.9.7	7.279	246.721	-27.28	.5682	8 147	264.8
14288.	.64687	. 5638	276.6	5.946	7.383	254.184	96*92-	.5557	65.651	265.2
14686.	.63122	.5676	277.0	5.945	7.491	261.594	-26.54	. 54.31	69.192	5.592
13686.	.6544	1265	277.4	5.94	7.682	269.197	-26.11	.5386	91.769	565.9
13508.	.65374	2265.	37.5	5.943	7.718	276.915	11.22-	.5179	94.385	2.992
13484.	.63239	.6113	278.2	5.941	7.635	264.753	-25.32	• 58 52	97.8-1	5.992
132 66.	.66725	. FBE7	278.5	5.9.0	7.963	292,716	16.42-	\$264.	99.736	566.3
13883.	.67145	.6114	279.1	5.939	7.978	3.9.695	-24.51	9524*	102.468	2.1.5

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LIFT AT OKOGUE PEL

DRAG AREA DROGUE

1834.35 66.

DRAG AREA HAIN CHUTE NT DROGUE AND HONRE

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24-17	20.53	24.87	25.18	25.46	25.74	26.82	26.28	15.92	26.88	27.85	27.38	27.54	27.78	28.42	28.25	28.48	28.78	26.82	29.14	98.62	29.57	29.78	29.99	39.20	30.41
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1187.76	11994,64	1614.96	1082-10	985.74	42.116	964.68	951.81	932.87	925.79		199.15		10.179	167.97	843.98						755.55	746.11	54.45	786.56	642.65
42.652	277.56	298-52	16.862	311.26	323.75	336. 4E	3.9.19	362.13	375.21	388.45	+81.85	11.511	\$17.624	443.83	11.754	471.34	485.77	516.39	12-515	538.53	545.45	550.89	576.35	532.44	:86.35
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Nomenclature

Program 14B

C1	=3.0413, used in equation of state (English units)
DAB	Drag area of the balloon, (CDSo) balloon
DAD	Drag area of the drogue, (CDSo)drogue
DAM	Drag area of the main chute, (CDSo) main
DAS	Drag area of the whole system, $(C_DS_o)_{system}$
DB	Drag of the balloon, DB
DD	Drag of the drogue, DD
DH	Increment of height, AH
DIAB	Diameter of the balloon, d _b
DM	Mass of helium transferred in time, Δt
DS	Drag of the system, D_S
DT	Increment of time, Δt , during which system drops a height ΔH
DTEMP	Increase in temperature, ΔT , of helium due to adiabatic heating from the balloon's dropping a height ΔH (200 ft), DTEMP=.8°/200 ft or 4°/1000 ft
F	Interpolation factor used to calculate last line of output
н	Height, H
HI	Initial height, H _i
IFLAG	Flag used to indicate whether calculations have been completed

NOMENCLATURE

I	Number of intervals of height from ground, i.e., $I = H/\Delta H$
II	Index for DØ loop
ITØP	Number of height intervals of the initial height from the ground, i.e., $IT \not DP = H_{1}/\Delta H$
J	Number of height intervals from ground in thousands of feet.
JJ	Number of height intervals from J to I
P	Pressure (P) at height H
PI	π
PRESS	Array containing values of pressure at 1000 ft. intervals, used in the calculation of P
Q	Dynamic pressure, $q = l_{10}v^2$
RLM	Lift to mass ratio (L/M) of helium in the balloon.
SIG	Array containing values of $\sigma = \rho/\rho_0$ at 1000 ft. intervals, used in the calculation of SIGMA
SIGMA	Ratio of σ = ρ/ρ at height H
т	Time, t
TAIR	Temperature of the air at height H
TEMP	Temperature of the helium in the balloon at height H
THE	Temperature of helium from the dewer
TL	Total lift at height H

TLREL

Total lift when drogue is to be

released.

TM

Total mass at height H

TMHE

Total mass of helium to be transferred

to balloon

TMM

Temporary storage for TM

TR

Transfer rate of helium

TREF

Temperature of air at H,

VE

Equilibrium velocity $(v_{e}(H_{i}))$ of system

VEAV

Average velocity over interval

VØL

Volume of the balloon

VO

Velocity at end of preceding interval

WD

Weight of the drogue and associated

hardware

WS

Weight of the system less the total lift

WSREF

Initial weight of the system

WW

- (WS - DS) and should always be very

close to zero

XJJ

Difference in height from H to next

lowest 1000-ft level

Adiabatic Hoating of the Gas in the Balloon

As the balloon descends into a thicker and thicker atmosphere, the balloon contracts due to the greater pressure causing the gas in the balloon to undergo a certain amount of adiabatic heating. This is known to be 4°K per 1000 ft of descent and is essentially constant throughout the region of interest.

In program P14B calculations done every 200 ft, and through this distance the gas (already in the balloon, not in the dewer) will heat up by .8°K. The temperature and transfer rate of the gas entering the balloon are assumed to be constant, and the gas in the balloon is assumed to be at all times in equilibrium, i.e., the temperature of the gas is uniform throughout the balloon. The relationship connecting the temperature, pressure, and volume of the gas is taken to be the equation of state for an ideal gas expressed as:

 $V = \frac{mT}{3.0413 P}$

where V is in cu ft, m in grams, T in °K, and P in atmospheres.

In each 200-ft interval the system drops a height of 200 ft in a time Δt , which is computed elsewhere in the program. During this time a mass Δm of helium at temperature T_{He} is transferred at a uniform transfer rate r, and we have:

 $\Delta m = r\Delta t$

Also, the mass m_0 of helium already in the balloon is heated .8° from its original temperature T_0 to a temperature of T_0 + .8°. The mass m of helium entering the balloon is at temperature T_{He} and mixes to form a new temperature T_{He} inside the balloon such that:

$$T = \frac{m_o(T + .8) + \Delta m T_{He}}{m_o + \Delta m}$$

The new volume of the balloon is:

$$V = \frac{(m + \Delta m)T}{3.0413 P}$$

where P is the pressure at the new height.

References

Carten, Andrew S. Jr., 1976, The Flight Test Aspects
 of the Air-Launched Balloon System (ALBS)
 Development Program, AFGL-TR-76-0196.

II. VOR Positioning Program

A. Introduction

The position of a free balloon after launch is found by tracking the balloon from two or more VOR stations which measure the direction of the balloon with respect to magnetic north. The position of the balloon is then found by plotting these bearings on a map of that region.

A program has been written for the HP-9810A programmable calculator which will calculate the balloon position from the station locations and their bearings on the balloon. Calculations are based on spherical geometry, which is exact (for a perfectly spherical earth), and, therefore, errors due to the plotting on the map and to distortions in the projection of the map are eliminated.

The mathematics are described in Section B.

The program stores the coordinates of as many as 8 VOR stations at one time. In this way the path of the balloon can be followed over a longer distance without having to change the station coordinates. The program works as follows:

After the latitudes, longitudes, and declinations of the 8 stations have been input, the program will then request magnetic bearings from any three of those stations.

The user will enter the number of each station he has

chosen and its corresponding bearing. Taking the stations two at a time, the program will then calculate intersection points among the three stations selected.

After the coordinates of these intersection points have been calculated and printed, the program will request a new set of bearings. These new bearings need not necessarily be from the same stations previously chosen; any 3 of the 8 may be selected.

A complete description on how to use the program is given in Section C.

B. Mathematics

1. Introduction

The projection on the earth of the line of sight from a VOR station to the balloon follows a great-circle path. Therefore, the balloon position can be found by calculating the location of the point of intersection of the two great circles which correspond to the lines of sight from fixes from the two given VOR stations.

Inputs to the calculation are the latitudes, longitudes, and declinations of the two stations and the directions of the balloon from the stations with respect to magnetic north. The declination is used to change the magnetic direction to a compass heading with respect to true north.

For each station, these quantities are used to form

one vector which points from the station to the balloon, and another vector which points from the center of the earth to the station. Taking the cross product of these two vectors produces a vector which is perpendicular to the plane of that great circle which is the projection of the line of sight from that station to the balloon. In other words, one set of station coordinates and a bearing from that station produces a vector which is normal to the plane of that great circle which corresponds to the line of sight from that station to the balloon.

Taking the cross product of two of these normal vectors produces a vector which points from the center of the earth towards the point of intersection of the corresponding two great circles, i.e., it points along the line of intersection of the two great-circle planes. From this vector one can compute the latitude and longitude of the balloon.

2. Input and Output

The input to the calculation is the latitudes, longitudes, declinations, and magnetic bearings of the balloon of two stations.

The latitudes are north latitudes.

The longitudes are input to and output from the program as west longitudes. However, for mathematical convenience, within the program and in the calculations which follow,

east longitude is used in order to be consistent with a right-handed Cartesian coordinate system.

In the program the magnetic bearings are converted to bearings with respect to true north by adding to each its corresponding declination (expressed as positive for an easterly declination).

The north latitude and west longitude of the balloon position are printed by the program.

3. Nomenclature

0	is the center of the earth
A, C	are two VOR stations
В	is the position of the balloon
θ A , θ B , θ C	are the north latitudes of positions A, B, and C, respectively
^λ A, ^λ B, ^λ C	are the east longitudes of positions A, B, and C, respectively
σλ, σΒ, σC	are unit vectors from 0 to A, B, and C, respectively. These vectors are local verticals
AN, CN	are unit vectors at A and B, respectively, which point due north, i.e., these vectors are local horizontals pointing north
ĀĒ, CĒ	are unit vectors at A and B, respectively, which point due east. i.e., local horizontals pointing east

 n_A, n_B are bearings on the balloon with respect to true north from stations A and C, respectively

AB, CB are unit vectors which point from A and C, respectively, along those great-circle paths which are in the direction of the balloon B, i.e., local horizontals pointing in the direction of the balloon

 \vec{n}_A, \vec{n}_B are vectors which are normal to the great-circle planes specified by the vectors \vec{AB} and \vec{CB} , respectively

4. Equations

a. The first step is to form the vectors \overrightarrow{OA} , \overrightarrow{AN} , and \overrightarrow{AE} for station A. The vectors \overrightarrow{OC} , \overrightarrow{CN} , and \overrightarrow{CE} will then be constructed in a similar fashion. The vectors are expressed in terms of their x, y, and z components respectively. The origin of this Cartesian coordinate system lies at the center of the earth; the z-axis points towards the north pole; the x and y axes lie in the plane of the equator and point toward the 0° and 90° meridians, respectively.

The x, y, and z components of these vectors can be expressed in terms of the latitude $\theta_{\mbox{$A$}}$ and longitude λ_A of station A.

Looking at Figures 1 and 2, one can see that the x, y, and z components, respectively, of these vectors are:

 $\overrightarrow{OA} = (\cos\theta_A \cos\lambda_A, \cos\theta_A \sin\lambda_A, \sin\theta_A)$

 $\overrightarrow{AN} = (-\sin\theta_A \cos\lambda_A, -\sin\theta_A \sin\lambda_A, \cos\theta_A)$ $\overrightarrow{AE} = (-\sin\lambda_A, \cos\lambda_A, 0)$

The vectors \overrightarrow{OC} , \overrightarrow{CN} , and \overrightarrow{CE} are constructed similarly, and their components can be found by substituting θ_C for θ_A and λ_C for λ_A in the vectors given above.

b. The next step is to form the vector \overrightarrow{AB} which points from A along the great-circle path which is the projection of the line of sight from A to B. In Figure 3 is a diagram of the local horizontal plane at A. From this we see that:

 $\overrightarrow{AB} = \overrightarrow{AN} \cos n_A + \overrightarrow{AE} \sin n_A$ Expanding this to find the x, y, and z components of \overrightarrow{AB} , we have:

 $\overrightarrow{AB} = (-\sin\theta_A \cos\lambda_A \cos\eta_A - \sin\lambda_A \sin\eta_A),$ $-\sin\theta_A \sin\lambda_A \cos\eta_A + \cos\lambda_A \sin\eta_A),$ $\cos\theta_A \cos\eta_A)$

The vector \overrightarrow{CB} is constructed similiarly, and its components can be found by substituting θ_C for θ_A , λ_C for λ_A and η_C for η_A in the above equations.

The next step is to form the vector \overrightarrow{n}_A which is normal to that great-circle plane which contains the line of sight from station A to balloon B. This vector can be formed by crossing any two vectors which lie in the plane (and are not parallel). The vectors \overrightarrow{OA} and \overrightarrow{AB} lie in this plane, and we have:

$$\overrightarrow{n}_{A} = \overrightarrow{OA} \times \overrightarrow{AB}$$

$$(\overrightarrow{n}_{A})_{X} = (\overrightarrow{OA})_{y} (\overrightarrow{AB})_{z} - (\overrightarrow{OA})_{z} (\overrightarrow{AB})_{y}$$

$$= \sin \lambda_{A} \cos \eta_{A} - \sin \theta_{A} \cos \lambda_{A} \sin \eta_{A}$$

$$(\overrightarrow{n}_{A})_{y} = (\overrightarrow{OA})_{z} (\overrightarrow{AB})_{X} - (\overrightarrow{OA})_{X} (\overrightarrow{AB})_{z}$$

$$= -\cos \lambda_{A} \cos \eta_{A} - \sin \theta_{A} \sin \lambda_{A} \sin \eta_{A}$$

$$(\overrightarrow{n}_{A})_{z} = (\overrightarrow{OA})_{X} (\overrightarrow{AB})_{y} - (\overrightarrow{OA})_{y} (\overrightarrow{AB})_{X}$$

$$= \cos \theta_{A} \sin \eta_{A}$$

The vector \vec{n}_B^{\star} is constructed similiarly, and its components can be found by substituting θ_C for θ_A , λ_C for λ_A and η_C for η_A in the above equations.

d. Taking the cross product \vec{V} of the vectors \vec{n}_A and \vec{n}_B produces a vector which points along the line of intersection of the two great-circle planes. This line of intersection passes through the center of the earth and the balloon. The vector formed will point either from the center of the earth to the balloon towards a north latitude or in the opposite direction towards a south latitude. The components of \vec{V} are:

$$v_{x} = (\overrightarrow{n}_{A})_{y} (\overrightarrow{n}_{B})_{z} - (\overrightarrow{n}_{A})_{z} (\overrightarrow{n}_{B})_{y}$$

$$v_{y} = (\overrightarrow{n}_{A})_{z} (\overrightarrow{n}_{B})_{x} - (\overrightarrow{n}_{A})_{x} (\overrightarrow{n}_{B})_{z}$$

$$v_{z} = (\overrightarrow{n}_{A})_{x} (\overrightarrow{n}_{B})_{y} - (\overrightarrow{n}_{A})_{z} (\overrightarrow{n}_{B})_{x}$$

In the case that \overrightarrow{v} points towards a south latitude, we can turn the vector around by making the transformation $\overrightarrow{v} \to -\overrightarrow{v}$.

e. Even though \overrightarrow{v} now points in the direction of \overrightarrow{OB} , it is not necessarily a unit vector. To form the unit vector,

v must be divided by its length and we have:

$$\overrightarrow{OB} = \overrightarrow{v} / |\overrightarrow{v}|$$

where $|\overrightarrow{v}| = \sqrt{v_x^2 + v_y^2 + v_z^2}$

6) Looking back at Figures 1 and 2, the coordinates of the balloon θ_B and λ_B can be found from \overrightarrow{OB} or \overrightarrow{V} by:

$$a_{B} = \sin^{-1}[(\overrightarrow{OB})_{z}] = \sin^{-1}(\frac{v_{z}}{|\overrightarrow{v}|})$$

$$\lambda_{B} = \cos^{-1}[\frac{(\overrightarrow{OB})_{x}}{\cos\theta_{B}}] = \cos^{-1}[\frac{v_{x}}{|\overrightarrow{v}|\cos\theta_{B}}]$$

C. The Program

1. Introduction

The following is a step-by-step description of how to use the VOR positioning program. A sample output, a listing, a flow chart, and an explanation of the organization of the memory are also provided in Figures 4 - 7, respectively. Supplementary information about the program is given which enables the user, when data has been entered incorrectly, to correct the mistake without having to go back to the beginning of the program.

2. How to Use the Program

Press END, CONTINUE.

The program then asks for the number of stations to be put into the calculator. The program can handle up to 8 stations. The user puts this number into the x register and then presses CONTINUE.

The program then prints a station number, starting at station 1 and going up to the number of stations requested

in the previous step, and requests the latitude, longitude, and declination, respectively, of that station. For each of these variables, enter the number of degrees in z, minutes in y, and seconds in x, and then press CONTINUE.

Latitude is input as north latitude, longitude as west longitude, and declination is input as a positive quantity for an easterly declination and as a negative quantity for a westerly declination.

After reading these quantities for a particular station, the program goes on to the next station, or, if the last station has just been input, it goes on to the next step.

The program then asks for the time of day. This is done so that there will be a record of the time that the VOR readings were taken. The user puts the time into the x register in the following way: 14:45 is put in as 14.45, etc. The user then presses CONTINUE.

The program then asks for bearings from any three of the stations, called 1ST, 2ND, and 3RD stations, respectively. For each of the three stations to be selected, enter the number of the station required in the y register and its corresponding magnetic bearing, expressed in decimal degrees, in the x register.

The program then prints the coordinates of the balloon location calculated using the 1st and 2nd stations, then the 1st and 3rd stations, and finally the 2nd and 3rd stations. For each station pair the latitude and longitude

are printed in degrees, minutes, and seconds.

The program then asks for a new time and then new station numbers and bearings. Again, any three stations may be selected, and these need not be the same ones which were selected previously. After each set of bearings has been input, and the balloon positions calculated, the program goes back to ask for a new time and a new set of station numbers and bearings.

A sample output is shown in Figure 4. All variables which the user must supply have an asterisk placed beside them. Note that all variables input by the user are printed on the paper tape immediately after being entered.

3. Supplementary Information

In order to help the user understand (and possibly modify) the program, a listing of the program is given in Figure 5 and a flow chart in Figure 6.

In the case that, after the station data have been entered, it is discovered that a station coordinate was entered incorrectly, or it is necessary that another station be substituted or added, it is usually more convenient to be able to enter the information directly into the memory rather than having the program do it.

(For the program to do this requires that all of the

information be reentered.) This requires a better knowledge of the memory and program organization. The organization of the memory is shown in Figure 7.

To put a number into a particular memory location, put the number into the x register, press x+(XTO) followed by the three-digit memory location. Since the coordinates entered in this program are always expressed in terms of degrees, minutes, and seconds, the user must first change this to decimal degrees before entering it into memory. In this case, put the numbers of degrees, minutes, and seconds into the z, y, and x registers, respectively, press K, 6 to convert to decimal degrees in the x register, and then press x+, followed by the three-digit memory address.

For station number i, i = 1 to 8, the latitude θ , longitude λ , and declination n are put into memory locations 5i+5, 5i+6, and 5i+7, respectively. The number of stations which are being stored in the memory is kept in memory address 000.

If a bearing is entered incorrectly, press GTO, 0, 2, 5, 0, and CONTINUE. The program will then ask for a time followed by three new station numbers and their corresponding bearings. The old station numbers and bearing are replaced.

D. Comparison of VOR Data with Radar Data

Using data from a particular balloon flight for which both VOR and radar data were available, a comparison of results using these two sets of data was made. Radar data are considered to produce substantially more accurate results than VOR data and, therefore, are suitable for determining the accuracy of the VOR data.

Errors in VOR positioning can result from a number of sources: 1) the error (Δη) in the magnetic bearing (η) of the target as viewed by the VOR station in question, 2) interference from other radio signals or mountains, and 3) the line of sight being too close to the horizon. The error in the bearing is the only type of error which will be discussed here. Errors due to the other two sources are highly unpredictable and would probably have to be measured empirically.

The error in positioning (E) due to one VOR station alone at a distance d from the target, assuming the bearing of the other station used for the fix is exact, is: $E = d \tan(\Delta n)$, or for small Δn , $E = d(\Delta n)$. The maximum error in a VOR bearing is known to be $\pm 1/2^{\circ}$. Errors $(\overrightarrow{E_1} \text{ and } \overrightarrow{E_2})$, due to the errors in the bearings of both of the VOR stations used to produce a given fix, add vectorially to produce the total error $\overrightarrow{E} = \overrightarrow{E_1} + \overrightarrow{E_2}$. This is shown in Figure 8.

When the angle α , the angle between the lines of sight from two given VOR stations, approaches 180° , i.e., when the balloon is very close to the line joining the two VOR stations, the total error $(\overrightarrow{E}_1 + \overrightarrow{E}_2)$ can become very large even though the bearings themselves are accurate, i.e., Δn_1 and Δn_2 are small. Conversely, an intersection angle of 90° between the lines of sight is optimum. This is shown in Figure 9. The shaded areas show where the balloon might be for each of the two fixes A and B, assuming an error of $\pm 1/2^{\circ}$ in each of the bearing-angle measurements. The centers of the shaded areas are the positions where the VOR predicts that the balloon is located. Note that for position B the fact that α is close to 180° can lead to a large error in the prediction of the balloon position.

Both radar and VOR data were available on balloon flight
H75-4, which took place on February 7, 1975. VOR data were
available from the following stations: 1) Truth or Consequences,

- 2) Deming, 3) Newman, Texas, 4) Holloman, 5) Pinon,
- 6) Hudspeth, Texas, 7) Roswell, and 8) Pecos, Texas. These stations and the position of the radar are shown on a map of the area in Figure 10. Balloon positions and their corresponding times are shown throughout the flight.

During this flight balloon positions were computed using Truth or Consequences, Holloman, Pinon, and Roswell. The other stations were too far away to provide accurate results.

Since E=d tan $\Delta\eta$, the error is proportional to the distance of the station from the target. Furthermore, the farther away the station is from the balloon, the closer to the horizon the balloon appears.

Larger discrepancies (2 to 4 nm) occurred at the beginning of the flight (before 9:14) when the balloon had not yet reached flight altitude and was therefore too close to the horizon to get a good fix. The radar was much closer to the balloon during this time, and its results are probably accurate.

Later in the flight (10:35 and after), discrepancies of 2 to 5 nm were seen. At this point the balloon was about halfway between Roswell and Pinon, so that a was close to 180°. Therefore, these two stations could not be used together to get a good fix. Holloman and Truth or Consequences were at this time quite far away (60 and 120 nm, respectively) and were also looking over a mountain range. Hence, using either one of these with either Roswell or Pinon also tended to produce larger errors than usual because of both the distance, increasing the error and making the balloon appear close to the horizon, and the interference produced by the mountains. This seems to have been in fact the case, because the fixes using Holloman were consistently about 3 nm to the south, whereas the fixes using Truth or Consequences were consistently about 3 nm to the north. Note

that the radar data could also have been influenced by the mountains and the balloon's being too close to the horizon. However, the fact that Holloman predicted positions were consistently 6 nm south of those predicted by Truth or Consequences shows that the VOR data themselves were inconsistent. During this part of the flight Carlsbad could have provided a good fix, but data from it were not available.

LIST OF FIGURES

- CROSS SECTION OF THE EARTH CONTAINING POINTS O,A, AND THE NORTH POLE
- 2. PROJECTIONS OF AN, AE, AND OA IN THE PLANE OF THE EQUATOR
- 3. VECTORS IN THE PLANE TANGENT TO THE EARTH AT A
- 4. SAMPLE OUTPUT
- 5. LISTINGS OF VOR PROGRAM
- 6. FLOW CHART OF VOR PROGRAM
- 7. ORGANIZATION OF MEMORY USED FOR VOR PROGRAM
- 8. ERROR IN BALLOON POSITION FROM VOR FIXES
- 9. EFFECT OF α ON THE ERROR IN BALLOON POSITION
- 10. MAP SHOWING BALLOON FLIGHT H75-4

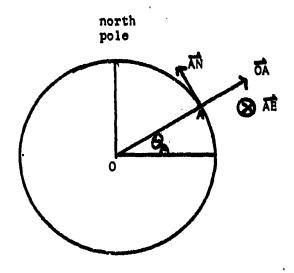


Figure 1. Cross Section of the Earth Containing Points O, A, and the North Pole.

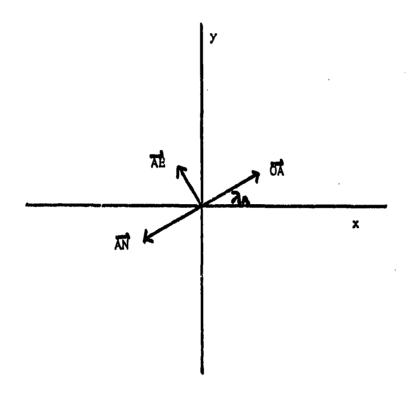


Figure 2. Projections of \overrightarrow{AN} , \overrightarrow{AB} , and \overrightarrow{OA} in the Plane of the Equator.

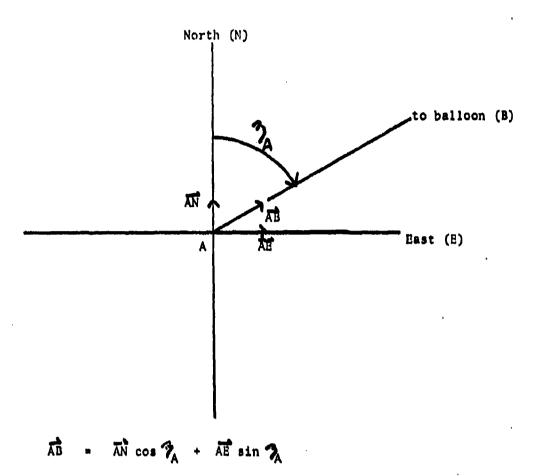


Figure 3. Vectors in the Plane Tangent to the Barth at A.

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Fig. 4 Sample output

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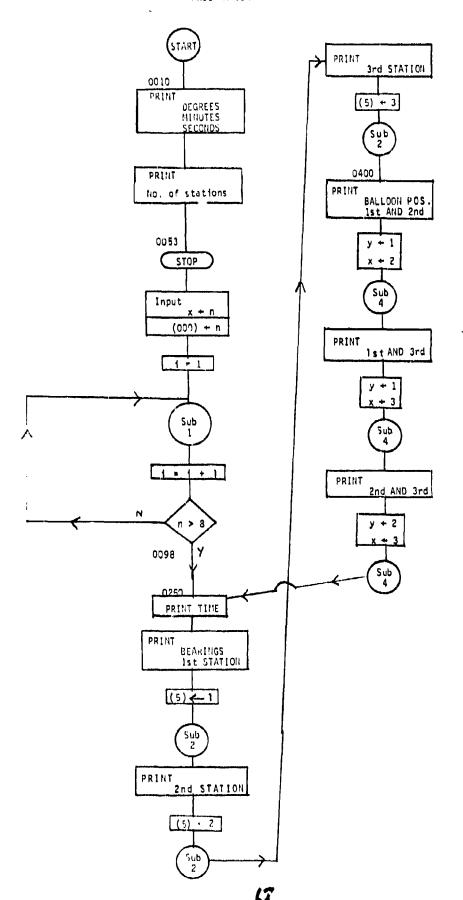
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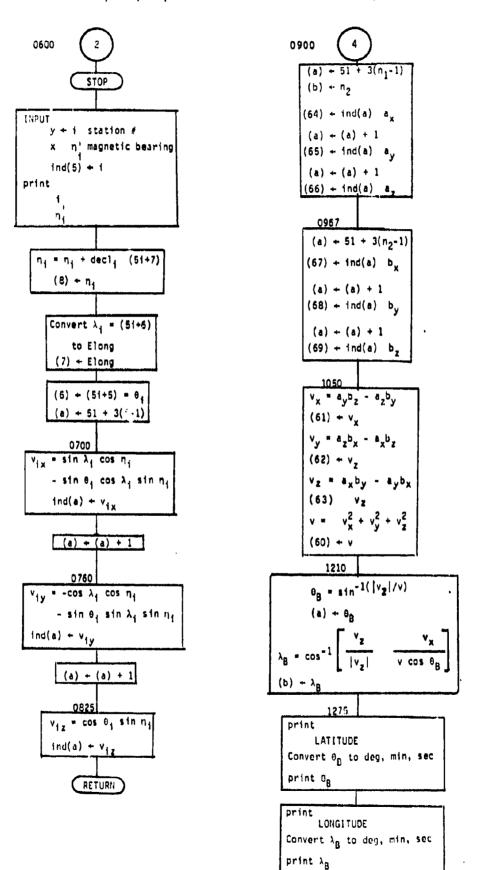
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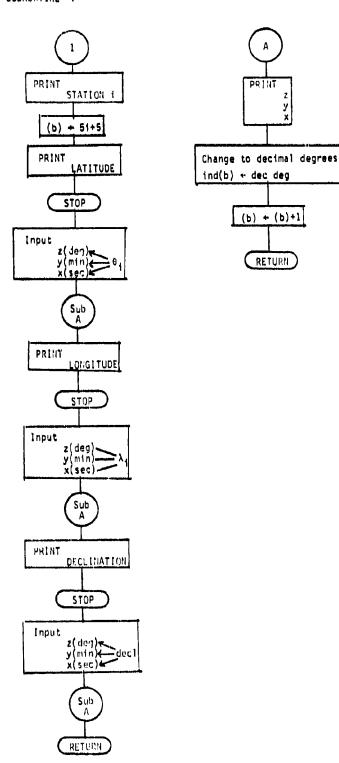
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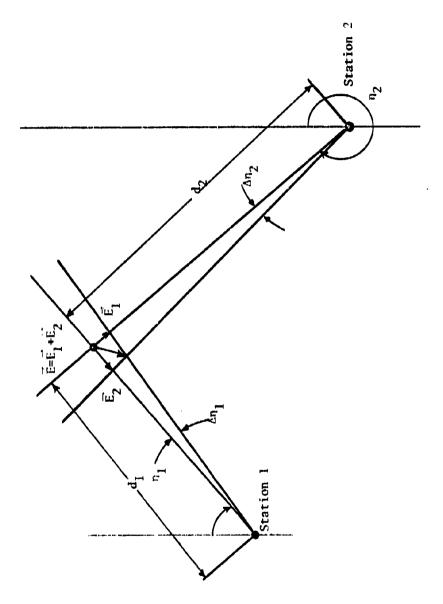




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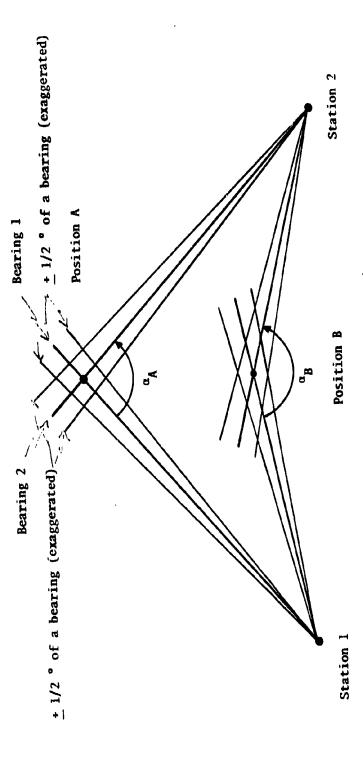
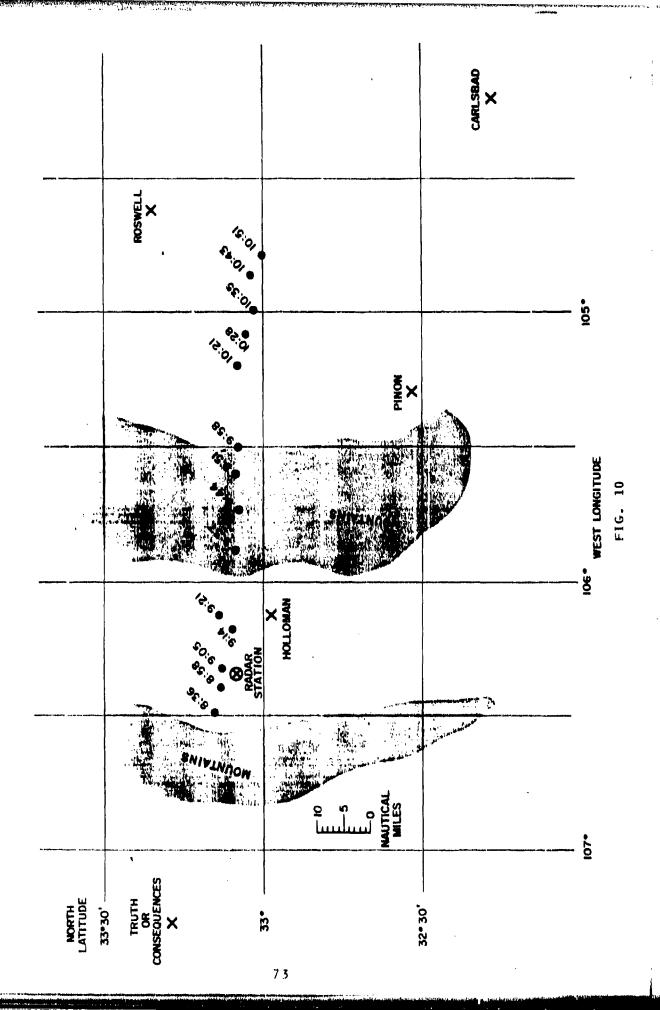


Fig. 9



III. Tethered-Balloon Experiment

Introduction

A series of tests was made in February-March 1975 to aid in the implementation of the Tactical Loran Tritether Instrumentation Hook-up'. These tests consisted of flying a 30000 cu ft balloon on tethers of various lengths (100, 500, 1000, and 2000 ft) with various loads (from no load up to 260 lbs) in various winds. The purpose of these tests was to determine the extent of balloon wander under different wind conditions. The balloon position was measured by means of theodolites, and wind azimuth and velocity were measured by instruments attached just under the balloon. Table 1 gives a summary of the different runs made in this series of tests.

Analysis

At the beginning the approach was to plot individual data points, with each graph containing data from only one run. Balloon positions with respect to ground zero were plotted with the y-axis representing north and the x-axis east. Values of wind direction and azimuth corresponding to each balloon position were also plotted on these graphs, with the distance from the origin corresponding to the wind velocity and the angle with respect to the y-axis corresponding to the wind azimuth. The two sets

of points were numbered and connected by straight lines, providing a kind of time history of the balloon position, wind direction, and wind speed.

Using the corresponding values of wind azimuth, the balloon positions were then resolved into components parallel and perpendicular to the wind. These values were also plotted as a function of time.

Since not much could be inferred from those types of plots, since the wind data might not have been good enough to justify a point-to-point breakdown, and since it takes a certain amount of time for a given change in wind direction or speed to effect a change in balloon position, it was decided to group together those times when both wind speed and direction were 'approximately constant.' Balloon positions during a given time period were resolved into components parallel and perpendicular to the wind based on the average azimuth of the wind over the whole period. Each time period contains at least five points, and, for some runs in which the wind was nearly constant, it was not necessary to divide the data at all. Balloon positions parallel and perpendicular to the wind were graphed versus time, and each run was graphed separately.

Since this type of time history of balloon wander did not seem to provide any notable insights, data from each of these time periods were grouped together, and statistics of mean, standard deviation, maximum excursion from ground zero, and spread of excursions about ground zero were calculated. These are statistics of the balloon displacement from ground zero both parallel and perpendicular to the wind. Hence, for each time period of a given run one value for each of the above statistics of balloon displacement was calculated, both parallel and perpendicular to the wind.

In the next set of graphs, each of the statistics of displacement listed above was plotted versus the average wind velocity of the corresponding time interval. Each graph contained the data from all runs of a given tether length, for a given statistical parameter either parallel or perpendicular to the wind. If a particular run was divided into more than one time interval, more than one set of statistics were plotted for that run.

Note that in the case of maximum and minimum excursions, the length of the time interval of the run is important; however, no attempt was made to take this into consideration. On the other hand, in the case of the mean and standard deviation, the length of the time interval is not important as long as it is 'long enough' to provide a representative sample.

From this set of graphs it was seen that in the runs of the 100' tether the balloon wander is a much higher fraction of the cable length than with the other tether lengths. Note that, in the application for which this will be used, it is the zenith angle of the balloon tether which is important. From this set of graphs it was decided that the 100' tether was not suitable.

One can also see that the 1000' and 2000' tethers are noticeably better than the 500' tether. However, it was felt that performance of these lengths would not justify their use over the 500' tether. Use of a longer cable increases the weight, thereby decreasing the net lift of the balloon, and makes the rigging of the balloon more difficult

Since it is the zenith angle of the balloon tether which is important in this application, the statistics of balloon displacement listed above were converted to percentage displacement (=displacement/cable longth x 100). Each statistic was plotted separately versus average wind velocity. For a given statistic, the data for the 500', 1000', and 2000' tothers parallel or perpendicular to the wind were plotted on the same graph.

For mean displacement parallel to the wind, the percentage displacement increases with wind velocity in a way which is almost linear. There is no noticeable dependence on tether length apart from the fact that the 500° tether does not perform quite as we'l as the two longer tethers. There is a fair amount of scatter in the graph, but the slope of that line which produces the best fit is 1.7% of cable length per

mph of average wind velocity. Therefore, the degree to which the balloon is blown back into the wind is proportional to the average wind velocity.

Mean displacement perpendicular to the wind showed no systematic variation with wind velcoity. This is reasonable since the balloon should not have a preferred average position perpendicular to the wind.

Standard deviaiton parallel and perpendicular to the wind is very variable and, with the exception of windsbelow 5 mph, shows no variation with wind velocity. Furthermore, the 500' tether data have noticeably higher standard deviations than do the 1000'- or 2000'- tether data. Note that even in high winds, greater than 20 mph, the standard deviation did not increase.

The maximum excursion from ground zero parallel to the wind showed an almost linear increase with wind velocity, but with considerable spread. Again the 500' tether shows, in general, greater displacements than do the two longer tether lengths.

The spread of excursions parallel to the wind shows no relation to wind velocity above 5 mph. But, here again, the 500' tother shows greater displacements.

The variation of these two statistics perpendicular to the wind also showed no variation with wind velcoity above 5 mph, but the 500' tether showed significantly greater

displacements than did the two longer tether lengths.

We have seen that parallel to the wind the mear and maximum excursion varied with wind velocity in an almost linear fashion. Dividing each of these statistics by wind velocity, (producing a mean or maximum displacement /mph average wind velocity), these data were plotted versus load to try to find what effect, if any, the load might have on balloon wander. Qualitatively, we know that an increase in the load will increase the angle of attack of the balloon. Up to a certain point this is desirable since it produces an increase in aerodynamic life, but after this point the aerodynamic lift decreases with angle of attack.

For the "mean displacement parallel to the wind/mph average wind velocity", there seemed to be some correlation with load, i.e., angle of attack. The spread is very large, but one can conclude that the slope of the percentage mean displacement parallel to the wind versus wind velocity is higher when the load is greater.

The slopes computed using the percentage-maximum-excursion statistics show, on the other hand, no correlation with load.

Summary

1) Data were first resolved into components parallel and perpendicular to the wind.

- 2) Data were then grouped together into time periods when both the speed and azimuth of the wind were constant.
- 3) Data for each time period during which the wind was constant were then grouped together to form statistics of the displacement of the balloon with respect to ground zero both parallel and perpendicular to the wind.
- 4) The percentage displacement of the 100' tether was too high to be suitable for this application. The percentage displacements for the 1000' and 2000' tethers were noticeably smaller than those of the 500', but their performance was not that much better to justify their use over the shorter 500' tether.
- indivaried linearly with wind velocity, with a slope of 1.7%/mph, the balloon being blown back into the wind more and more as the wind increased. Perpendicular to the wind the mean displacement showed no variation with wind velocity, i.e., the balloon had no preferred position perpendicular to the wind.
- 6) Maximum excursion parallel to the wind varies
 linearly with wind velocity. This is due to a great extent,
 however, to the variation of the mean. Perpendicular to the
 wind there is no systematic variation.
- 7) Standard deviation parallel and perpendicular to the wind showed no variation with the wind above 5 mph.

- 8) In the case of the standard deviation parallel and perpendicular to the wind and of the maximum excursion parallel to the wind, the 500' tether showed greater percentage displacements, in general, than did the two longer tethers.
- 9) Load, in general, seems to produce greater displacements parallel to the wind.

Table 1

	NO LUAD	LOAD BAR ONLY	LOAD BAR +140 LOS BALLAST	LOAD BAR +265 LBS BALLAST
2000' tether	2/26 #4 2/27 #4 3/4 #4*	3/7 #2* 3/7 #3	2/27 #8 2/28 #4 3/6 #4	3/7 #1
1000' tether	2/26 #3 2/27 #3 3/4 #3	3/7 #4*	2/27 #7 2/28 #3 3/6 #3	3/3 #3
500' tether	2/26 #2 2/27 #2 3/4 #2	2/26 #5 3/3 #5	2/27 #6 2/28 #2 3/6 #2	3/3 #2* 3/7 #6
100 tether	2/26 #1 2/27 #1 3/4 #1	3/3 #4	2/27 #5 2/28 #1 3/4 #5 3/6 #1	3/3 #1* 3/7 #5

NO WIND DATA

IV Balloon-Flight Simulation

The following balloon-flight simulations were done.

A. Eglin AFB simulation

Using 10 years of wind data from Eglin AFB, balloon positions and statistics of balloon positions were calculated assuming an ascent to 10 mb at a constant ascent rate of 800 ft/min, letting the balloon float for 0, 1, 4, 6, 8, and 10 hours at 10 mb. This was done using wind data for days 1-10, 11-20, and 21-31 in each of the months from April to September.

The wind data were received on a 9-track, phase-encoded tape which had to be converted to a 7-track tape in order to make it compatible with the CDC 6600 at AFGL. Before writing the 7-track, 800-bpi tape the data were sorted by month and day.

The balloon positions and associated statistics were done by a program called CHICSIM, which was modified for this particular job.

B. Holloman and White Sands simulation

Using 10 years of wind data from Holloman and White Sands, balloon positions and statistics of balloon positions were calculated assuming ascents to 80, 90, and 100 kft, at a constant ascent rate of 800 ft/min, letting the balloon float for 0, 2, 4, 6, 8, 10, and 12 hours at each of those altitudes.

Each of the above cases was run for days 1-15 and 16-31 for all months of the calendar year.

These wind data were already on a CDC 7-track tape.

The data were processed by a program called SIMBALL, which was modified slightly for this particular job.